



U.S. Department of Transportation
Federal Aviation Administration

FEDERAL AVIATION ADMINISTRATION – PARTS MANUFACTURER APPROVAL

Aerospace Welding Minneapolis, Inc.
1045 Gemini Road
Eagan, MN 55121

PMA No.: PQ2712CE
Supplement No.: 96
Date: March 8, 2018

Article Name	Article Number	Approved Replacement For Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Left Rear Stack Assembly	A62231-005	Piper Aircraft Inc. 62231-005	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A62231-005 <u>Rev:</u> - <u>Date:</u> 01/15/2018 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up
Right Rear Stack Assembly	A62232-004	Piper Aircraft Inc. 62232-004	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A62232-004 <u>Rev:</u> - <u>Date:</u> 01/15/2018 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up
TailPipe Assembly	A85477-011	Piper Aircraft Inc. 85477-011	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A85477-011 <u>Rev:</u> - <u>Date:</u> 07/20/2017 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up
Bolt Flange	A690-091	Piper Aircraft Inc. 690-091	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A690-091 <u>Rev:</u> - <u>Date:</u> 09/07/2017 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up
Tailpipe Hanger Clamp	A24791-006	Piper Aircraft Inc. 24791-006	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A24791-006 <u>Rev:</u> - <u>Date:</u> 01/09/2018 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up
Strap	A24793-004	Piper Aircraft Inc. 24793-004	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A24793-004 <u>Rev:</u> - <u>Date:</u> 01/09/2018 or later FAA-approved revisions	Piper Aircraft Inc.	PA-28-181 S/N 2843001 and up

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GENERAL NOTES:

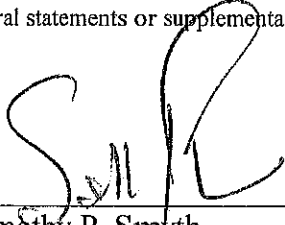
Provide minor design changes in a manner as determined by the ACO. Process major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.

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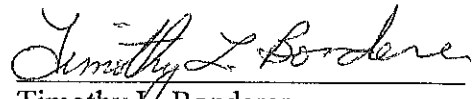
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The FAA accepted the ICA approach for the above articles with their designs. These ICA may refer to those of the respective articles from the holders of type certificates. Otherwise, provide supplementary ICA for differences in the replacement articles. Make referral statements or supplemental ICA readily available per 14 CFR § 21.50.



Timothy P. Smyth
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Timothy L. Bonderer
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