



U.S. Department of Transportation  
Federal Aviation Administration

FEDERAL AVIATION ADMINISTRATION – PARTS MANUFACTURER APPROVAL

Aerospace Welding Minneapolis, Inc.  
1045 Gemini Road  
Eagan, MN 55121

PMA No.: PQ2712CE  
Supplement No.: 8 0  
Date: FEB 12 2016

Article Name	Article Number	Approved Replacement For Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Tailpipe	A1250860-164	1250860-6, 1250860-97, 1250860-110, 1250860-164	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-164 <u>Rev:</u> A <u>Date:</u> 11/24/2015 or later FAA-approved revisions	Textron Aviation Inc.	TP206D S/N P206-0575 thru P206-0603, TP206E, TU206D S/N U206-1323 through U206-1444, TU206E, TU206F S/N U20601701 and on, T210F, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R
Tailpipe Adapter	A1250860-168	1250860-78, 1250860-168	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-168 <u>Rev:</u> A <u>Date:</u> 11/17/2015 or later FAA-approved revisions	Textron Aviation Inc.	TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F up to S/N U20602199, TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206F S/N U2060220 and on, TU206G S/N 676 and U20603522 and on, T207, T207A, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R
LR Stack Assy.	A1250860-121	1250860-5, 1250860-90, 1250860-121	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A01250860-121 <u>Rev:</u> A <u>Date:</u> 12/24/2015 or later FAA-approved revisions	Textron Aviation Inc.	TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F, TU206G S/N 676 and U20603522 and on, T207, T207A, T210F, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R

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Article Name	Article Number	Approved Replacement For Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Turbo Inlet	A1250860-122	1250860-4, 1250860-105, 1250860-122	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-122 <u>Rev:</u> A <u>Date:</u> 12/17/2015 or later FAA-approved revisions	Textron Aviation Inc.	TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F, TU206G S/N 676 and U20603522 and on, T207, T207A, T210F, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R
RF Stack Assy.	A1250860-120	1250860-3, 1250860-72, 1250860-120	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-120 <u>Rev:</u> A <u>Date:</u> 12/23/2015 or later FAA-approved revisions	Textron Aviation Inc.	TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F, TU206G S/N 676 and U20603522 and on, T210F, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R
LF Stack Assy.	A1250860-157	1250860-157, 1250860-203	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-157 <u>Rev:</u> A <u>Date:</u> 12/22/2015 or later FAA-approved revisions	Textron Aviation Inc.	TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F, TU206G S/N 676 and U20603522 and on, T207, T207A, T210H, T210J, T210K, T210L, T210M, T210N, T210R
Rear Lower Turbo Support	A1250818-200	1250818-200	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250818-200 <u>Rev:</u> A <u>Date:</u> 12/23/2015 or later FAA-approved revisions	Textron Aviation Inc.	TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F up to S/N U20602199, TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206F S/N U2060220 and on, TU206G S/N 676 and U20603522 and on, T207, T207A, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R

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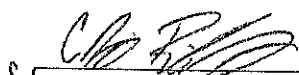
<u>Article Name</u>	<u>Article Number</u>	<u>Approved Replacement For Article Number</u>	<u>Approval Basis and Approved Design Data</u>	<u>Make/TCH Eligibility</u>	<u>Model/Series Eligibility</u>
Front Lower Turbo Support	A1250818-201	1250818-201	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250818-201 <u>Rev:</u> A <u>Date:</u> 12/23/2015 or later FAA-approved revisions	Textron Aviation Inc.	TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F up to S/N U20602199, TP206A, TP206B, TP206C, TP206D, TP206E S/N P206-0001 and P20600604 through P20600647, TU206F S/N U2060220 and on, TU206G S/N 676 and U20603522 and on, T207, T207A, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R
Clamp	A1250860-29	1250860-29	Test and Computations per 14 CFR § 21.303 <u>DWG No:</u> A1250860-29 <u>Rev:</u> C <u>Date:</u> 9/23/2015 or later FAA-approved revisions	Textron Aviation Inc.	P206 S/N P206-0001 and P20600604 through P20600647, TU206, TU206A, TU206B, TU206C, TU206D, TU206E, TU206F up to S/N U20602199, T210G, T210H, T210J, T210K, T210L, T210M, T210N, T210R

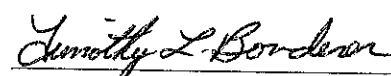
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GENERAL NOTES:

Provide minor design changes in a manner as determined by the ACO. Process major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.

The FAA accepted the ICA approach for the above articles with their designs. These ICA may refer to those of the respective articles from the holders of type certificates. Otherwise, provide supplementary ICA for differences in the replacement articles. Make referral statements or supplemental ICA readily available per 14 CFR § 21.50.

  
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